

Introduction to “To the Question of Energy Use of Detonation Combustion” by Ya. B. Zel’dovich

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YA. B. Zel’dovich (1914–1987) made numerous contributions¹ to the theory of detonation, beginning with his very well-known and widely translated article² on detonation structure that first introduced the standard Zel’dovich-von Neumann-Döring (ZND) model of shock-induced combustion. Even at that early stage of detonation research, Zel’dovich was also considering the application of detonations to propulsion and power engineering. He published these ideas in another paper³ that has been virtually unknown in the West and has apparently remained untranslated until now. We are indebted to Sergey Frolov of the N. N. Semenov Institute of Chemical Physics for first bringing this article to our attention. We believe that the focus of this paper, which is the application of detonation waves to power generation and propulsion, is very relevant to the current activity on pulse detonation engines. In particular, Zel’dovich was apparently the first researcher to consider the questions of the relative efficiency of various combustion modes, the role of entropy production in jet propulsion, and the distinction between unsteady and steady modes of detonation in power engineering and propulsion applications. Even 60 years later, we believe that his results are relevant and can be of value in modern discussions on thermodynamic cycle analysis of detonation waves for propulsion.⁴ For these reasons, we have arranged for the paper to be translated and suggested that it be published by the *Journal of Propulsion and Power*.

The paper is clearly written, and there is no need for extensive commentary, so we only sketch some connections with contemporary work. Sections 1–3 are concerned with the correct computation of the energy budget in an unsteady cyclic process and the thermodynamic efficiency. Zel’dovich recognizes that one has to account for the work necessary to sustain the detonation wave (through a piston, for example) when calculating the work that can be done by the products. This idea was also independently developed by Jacobs⁵ and later Fickett and Davis,⁶ although they were concerned primarily with high explosives. More recently, we have revisited this idea⁷ and carried out computations for mixtures and conditions relevant to pulse-detonation-engine operation. To our knowledge, Zel’dovich was the first researcher to conduct a thermodynamic analysis of a cycle involving a detonation. His conclusion that the efficiency of this cycle is always slightly larger than that of a cycle using constant-volume combustion (Humphrey cycle) has been confirmed many times since.^{8–10} Zel’dovich’s formal results for the thermal efficiency are identical to the results of recent studies.^{7,9} The specific numerical values given in the main body are, as Zel’dovich recognized, rough estimates and deviate substantially from detailed computations based on realistic thermodynamic properties. Despite the incorrect values for thermodynamic states, his final results regarding the differences in cycle efficiency are quantitatively correct. Repeating⁷ his computations with realistic thermodynamic properties gives a value of $\eta'_B = 0.26$ and $\eta_D = 0.30$ for the C_2H_4 -air exam-

ple discussed in Sec. 3, which yields the same 13% increase in the efficiency of the detonation cycle over constant volume as Zel’dovich estimated. Clearly, Zel’dovich knew that his results could be open to criticism because of the roughness of his estimates, and he addressed this with his late addition of the final section “Note Added in Proof.” Those results are within 1% of values computed with modern values of thermodynamic properties and numerical solution of the equilibrium states.

In Sec. 4, Zel’dovich considered using a detonation wave in a steady-flow airbreathing engine. Looking at a detonation wave as a shock wave followed by a reaction zone, he qualitatively argued that this process generates more entropy than a deflagration and showed that using a steady detonation instead of a deflagration resulted in a lower thrust, in agreement with many later studies.^{11–14} He gives a numeric example for a very simplified situation [ramjet traveling at the Chapman–Jouguet (CJ) velocity], for which the thrust of a detonation-based ramjet is a factor of two lower than that of an ideal isentropic inlet with constant-pressure combustion. As in the cycle analysis, his numerical values of the exhaust velocity are only rough estimates, and results using realistic thermochemistry yield values of Δu that are approximately a factor of two higher than Zel’dovich estimated. Reevaluation using realistic thermodynamic properties gives a value corresponding to Fig. 4 of $\Delta u = 438$ m/s, and that corresponding to Fig. 5 is $\Delta u = 950$ m/s. The ratio of the thrust for the constant-pressure case to the detonation case is 2.2, exactly the same as Zel’dovich found so that his final conclusions are not only qualitatively but also quantitatively correct.

Although Zel’dovich correctly concludes that the performance of steady detonation-based engines is inferior because of the irreversible entropy generation in the shock wave, he makes no attempt to reconcile this with the minimum entropy character of the CJ state that he discussed in Sec. 3 and subsequent authors have taken as the formal basis for the superiority of detonation-based power generation or propulsion. Recently, we have reexamined⁷ this issue and shown that the difference between the constraints in upstream states for steady (fixed stagnation conditions) and unsteady applications (fixed static states) is key in resolving this apparent contradiction.

The notation and units used by Zel’dovich are reasonably clear. The energy units are given in cal/mol for heat of combustion and enthalpy; the heat capacity units are cal/mol·K; the pressure units are kgf/cm², kgf/cm² = 0.980665 bar. In accord with the practice in chemical physics literature of that era, an explicit conversion factor between thermal and mechanical units is not used. Most symbols are defined in the text and have the usual modern meanings; the symbol J is used to denote the heat content, which present-day readers will recognize as the specific enthalpy. Although the reaction formula given in Sec. 4 does not include nitrogen, the numerical values and setting of the problem make it clear that Zel’dovich is considering the explosive mixture to be stoichiometric C_2H_4 air, and his initial conditions are 1 bar and 300 K. The figures have been redrawn and translated for clarity but are strictly faithful to the originals with the exception of the addition of axes labels to Fig. 3.

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References

- ¹Ostriker, J. P. (ed.), *Chemical Physics and Hydrodynamics*, Selected Works of Yakov Borisovich Zeldovich, Vol. I, Princeton Univ. Press, Princeton, NJ, 1992, pp. 411–458.
- ²Zel'dovich, Y. B., "On the Theory of the Propagation of Detonations in Gaseous Systems," *Journal of Experimental and Theoretical Physics*, Vol. 10, 1940, pp. 542–568; available in translation as NACA TM 1261, 1950.
- ³Zel'dovich, Y. B., "To the Question of Energy Use of Detonation Combustion," *Journal of Technical Physics*, Vol. 10, 1940, pp. 542–568 (in Russian).
- ⁴Roy, G. D., Forlov, S. M., Borisov, A. A., and Netzer, D. W., "Pulse Detonation Propulsion: Challenges, Current Status, and Future Perspective," *Prog—Energy Combustion Science*, Vol. 30, 2004, pp. 545–672.
- ⁵Jacobs, S. J., "The Energy of Detonation," U.S. Naval Ordnance Lab., NAVORD Rept. 4366, White Oak, MD; available as NTIS AD113271—Old Series.
- ⁶Fickett, W., and Davis, W. C., *Detonation Theory and Experiment*, Dover, Mineola, NY, 2001, Chap. 2, pp. 35–38.
- ⁷Wintenberger, E., and Shepherd, J. E., "Thermodynamic Analysis of Combustion Processes for Propulsion Systems," AIAA Paper 2004-1033, Jan. 2004.
- ⁸Kailasanath, K., "Review of Propulsion Applications of Detonation Waves," *AIAA Journal*, Vol. 38, No. 9, 2000, pp. 1698–1708.
- ⁹Heiser, W. H., and Pratt, D. T., "Thermodynamic Cycle Analysis of Pulse Detonation Engines," *Journal of Propulsion and Power*, Vol. 18, No. 1, 2002, pp. 68–76.
- ¹⁰Kentfield, J. A. C., "Thermodynamics of Airbreathing Pulse Detonation Engines," *Journal of Propulsion and Power*, Vol. 18, No. 6, 2002, pp. 1170–1175.
- ¹¹Dunlap, R., Brehm, R. L., and Nicholls, J. A., "A Preliminary Study of the Application of Steady-State Detonative Combustion to a Reaction Engine," *Jet Propulsion*, Vol. 28, No. 7, 1958, pp. 451–456.
- ¹²Foa, J. V., *Elements of Flight Propulsion*, Wiley, New York, 1960, Chap. 13, pp. 274–287.
- ¹³Sargent, W. H., and Gross, R. A., "Detonation Wave Hypersonic Ramjet," *ARS Journal*, Vol. 30, No. 6, 1960, pp. 543–549.
- ¹⁴Wintenberger, E., and Shepherd, J. E., "The Performance of Steady Detonation Engines," AIAA Paper 2003-0714, Jan. 2003.